

## Active Optics for UV/Vis/IR Space Telescopes

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# Active Optics for Space Telescopes



- Active Optics: Mirrors that can be reshaped after launch; and the Wavefront Sensing and Control system to command them
  - Reduce mission risk
    - Correct any optical problem that might arise
    - Enable testing to spec during system assembly and integration
  - Reduce mission cost
    - Reduce mission mass
    - Relax fabrication and assembly tolerances
    - Speed up Assembly, Integration and Test phases



# **Actuated Hybrid Mirrors (AHMs)**

#### • AHMs are large mirrors

PMs or PM segments

#### Nanolaminate facesheet

 Multilayer metal foil, made by sputter deposition on a super-polished mandrel

#### SiC substrate

 Reaction-bonded Ceraform SiC is cast in a mold, fired, then bonded to facesheet

#### Electroceramic actuators

 Surface-parallel embedded actuators give large stroke and high accuracy



- AHMs are low mass and high strength
  - Areal density < 20 kg/m<sup>2</sup> including electronics for meter-class AHMs
- AHMs are made by replication for high optical quality and low cost



# **Polished SiC Mirrors**

# Polished SiC mirrors are AHMs without a bonded Nanolaminate facesheet

- Nanolaminate facesheet
  - Multilayer metal foil, made by sputter deposition on a super-polished mandrel
- SiC substrate
  - Reaction-bonded Ceraform SiC is cast in a mold, fired, then bonded to facesheet

#### Electroceramic actuators

 Surface-parallel embedded actuators give large stroke and high accuracy



- Polished SiC mirrors are also low mass and high strength
- Polished SiC mirrors can be joined to create very large mirrors
- Polished SiC mirrors can be used at cold or even cryo temperatures



# **Nanolaminate Properties**

- Nanolaminates: multilayer solids with high interface concentration
  - Have been made from 72 materials
  - Amorphous/crystalline layers for AHM
- X-ray optic example has layer thicknesses from 0.4 nm to 32 nm





- AHM nanolaminate layers:
  - A few Å of C for release layer
  - Au layer for outer surface
  - Conventional coating applied
  - 446 periods of:
    - 42 nm crystalline Zr layer
    - 3 nm amorphous Zr/Cu layer
- Finished 1.52 m nanolaminate being removed from chamber



# **Ceraform Silicon Carbide**

#### Ceraform SiC:

- Fugitive core foam mold created by CNC machining
- SiC nanopowder slip fills mold
- Part is freeze-dried
- Mold core is leached out
- First firing creates green state part
- Part is machined
- Second firing to full hardness
- Final rough grind of SiC front surface matches the curvature of the mandrel/nanolaminate to ± 5 µm

Property	Units	Aluminum	Beryllium	SiC	ULE	Desire
ρ, Weight	g/cm3	2.71	1.85	2.95	2.21	Low
E, Stiffness	GPa	68.3	303	364	67.6	High
E/p,Specific Stiffness	KN-m/g	25	164	123	31	High
σ/ρ, Stress Loading	N-m/g	46	11	24	3.2	High
α,Thermal Soaks	ppm/°C	22.7	11.4	3.38	±0.03	Low
Δα Homogeneity	ppb/°C	100	100	30	10	Low
K/α, Thermal Gradients	MW/m	6.9	19	51	44	High
K/rCp,Thermal Diffusivity	m2/s	6.55	6.07	8.7	0.08	High
K/αE, Thermal Stress	MW-m/N	101	63	140	646	High





# **Joined SiC Mirrors**

- SiC substrates can be joined using brazing or bonding techniques, and then polished, to make very large, active mirrors
  - 4 m or larger, using existing SiC fab infrastructure
  - Directly polished to <20Å surface roughness</li>
  - Superpolishable to <5Å after Si cladding</li>
  - With or without central hole
  - Can be used at cold temperatures

# AHM and SiC Mirror Technology Status

- AHM mirror technologies are maturing rapidly
  - To do: grow to larger sizes (1.8m, 2.5m, e.g.)

#### Very large polished SiC mirrors offer benefits in some cases

- Large monolithic primary mirrors (4m or larger)
- Cold or cryogenic mirrors, active or not
- Very large polished SiC mirrors require some further technology development
  - To do: Lightweight mirror segment joining
  - To do: Low-stress Si cladding
  - To do: Superpolishing
  - To do: Cryogenic active mirrors, using actuators to correct cool-down stresses and avoid costly cryo-null figuring
- Other active optics technologies needing development
  - To do: "Self-sensing" for <10pm WFE stability</li>
  - To do: Continuous, pm accuracy WFS for internal coronagraph or lensing applications



# **Wavefront Sensing and Control**

- Wavefront sensing and control methods are well established
  - JWST
  - Active mirror testbeds
- Proposed exoplanet-specific WFSC methods need further development
  - Continuous pm-level WFS
  - Mirror self-sensing methods
- (Details in backup charts)

# **3-Dimensional Laser Truss**



- Uses Laser Distance Gauges (LDG)
  - 6 LDGs per segment measure all relative RB DOFs in the entire OTA
    - All PM segments, the SM, FF, TM and OBA
  - The IRS is attached to the OBA, providing measurements of 6 more absolute DOFs wrt inertial space
- Same measurement equation:  $\delta = Cx$ 
  - Sensitivities computed from model kinematics
- Measurement is invertible:  $x = C^{-1}\delta$  is full rank
- Optical State Estimator uses a Kalman Filter to estimate the RB state
  - Balances measurement vs. prior knowledge for optimal estimate
  - Predicts WF and Boresight from state estimate
- Feedback control using RB actuators and optimal control laws keeps performance in spec
  - Integrated model will be used to evaluate performance



## **Block Diagram**



- Major elements include
  - Wavefront Sensing and Control
  - Laser Truss Active Alignment: active WF compensation and LOS pointing control
  - Segment Thermal Control to stabilize optical figure
  - Isolation and Damping to attenuate vibration disturbances



#### Laser Truss Keeps All Optics Aligned



- Laser Truss measurements at high BW are processed in a Kalman Filter to estimate the perturbation state of all the optics
- Estimated state is fed back to control WFE at low BW and boresight at high BW



## **Laser Truss Pros and Cons**

#### • Pros

- High accuracy < 1 nm per LDG when  $\Delta$  angle is small
- Observes all important RB states including Primary and Secondary Mirrors, and Optical Bench
- Low drift with 1 laser feeding all LDGs, require WFS update once per day
- Light weight beam launchers
- No on-segment power dissipation
- Does not require segments to be close together
- Does not require any particular gap geometry
- Works with missing segments (no degradation for the segments that remain)
- Useful for I&T
- Degrades gracefully if individual LDGs go out
- Cons
  - Requires 12 fibers into each segment for 6 DOF



# Conclusion

- AHMs provide high-quality, low-mass large optics
  - Polished SiC mirrors promise the same advantages for very large mirrors, or cold mirrors
- Active optics compensate typical space telescope errors to reduce mission risk and cost
  - 10x to 300x for low-order errors, depending on actuator count
- Active optics relax fabrication and assembly tolerances system-wide, lowering cost
- Active optics permit testing to spec performance on the ground, at multiple stages of assembly, without complex GSE



#### BACKUP



#### **Actuators**

#### Sintered body



Electrical Connection (conceptual)



XiRE 0313 Photo, XiRE 0416 similar



Conductive polymer

Top surface: Conformal coating

NGX actuators use PMN-PT electrostrictive ceramics

100 - 200

 Multiple layers of ceramic and conductive electrode are co-fired to form a solid body

Active PMN Layer

Pt Electrode Layer

Thickness : 2-4 µm

# of active layers:

Thickness : 100 –152 µm

- Conductive polymers for external electrode and wire bonding (no soldering)
- Conformal insulating polymer coating

- High stroke, low voltage
  - ±2.5 um stroke at 20C
  - 0-100V operating range
- Used for astronomical Deformable Mirrors
  - High reliability

#### Actuator with Mounting Tabs





# **Nanolaminate Facesheet**

- AHM nanolaminates are made at LLNL, in the Very Large Optic Coater (VLOC)
- Mandrel is a nanoclean, superpolished glass tool with figure opposite to final AHM
- Mandrel is translated and rotated under "targets:" the deposition sources





- Magnetrons create Ar+ plasma to drive atoms off the targets and onto the mandrel
- Switching between multiple targets creates multilayers
- Nanolaminate uniformity and strength assured by: ultrastable processes
- Nanolaminate surface smoothness replicates mandrel

# Substrate Design Considerations



#### Substrate design must meet multiple objectives

- Optical performance is improved with more cells/actuators
  - At the expense of mass and complexity
  - Improved with multiple levels of ribs with differing heights
- Stiffness: first mode >> 100 Hz
- Mass: areal density typically 7-10 kg/m<sup>2</sup> for meter-class AHMs
- CTE balanced by selection of actuator interface tabs
- FEA models are built for candidate designs
- Structural/optical analyses are used to trade design objectives and constraints
  - Correctability
  - Mass
  - Stiffness
  - Actuator tab material





# **Wavefront Sensing and Control**



#### WFS&C Elements

- Imaging Camera ("PRC") with area array detector and narrow-band filter
- Focus adjust mechanism
- PM actuator electronics
- Shack-Hartmann Camera ("SHC"), with area array detector, pupil imaging lens and lenslet array

#### WFS&C Operations are performed while observing a star

- Initialization WFS&C uses SHC for large WF capture range ( > 30  $\lambda$  ), and PRC for high resolution and high accuracy
  - Use of PRC Imaging Camera measures WF in the main science camera no non-common path
  - Run once at the beginning of the mission
- Maintenance WFS&C uses PRC only, with minimal/no impact on science ops
  - Keeps WFE within spec
  - Run periodically throughout the mission (1/day to 1/week rate)

#### Image-based WF sensing using the PRC

Modified Gerchberg-Saxton (MGS) phase retrieval software proven through operations on many platforms



# **SHC Large Capture Example**

# SHC Image



 Data taken using white light source and open filter

WF Reconstructed from SHC Image

- WFS&C Experiment used an AHM, portable SHC and PRC, and autocollimating flat
- SHC results show large capture range WF control
  - Initial SHC WF error was 31 um (P-V), 6 um (RMS), double-pass
- After SHC control, WF error was 80 nm RMS in the SHC, 116 nm in the PRC





# **PRC Fine Control Example**



- WFS&C Experiment continues using the PRC imaging camera for imagebased WF sensing
- One or more iterations of control to achieve diffraction-limited WFE
- Performance is confirmed by the high-quality "single-pixel" in-focus PSF



10

20

30

10

0

20

30

40